

Section 1: Record of maintenance and elementary work

1

Date / /	Total air time	Total landings/cycles	Details of task	Signature	AME or AMO/ACA
2015/06/08	0.0	0/0	A/C Reg: C-GNVA Model: DHC-6 400 S/N: 925 TTSN 0.0 TCSN 0 Landings 0 L/H Engine S/N: PCE-RB0886 TSN: 0 CSN: 0 R/H Engine S/N: PCE-RB0887 TSN: 0 CSN: 0 L/H Prop S/N: BUA33359 TSN: 0 R/H Prop S/N: BUA33360 TSN: 0 June 8, 2015 This aircraft has been manufactured IAW Viking Air Limited Work Order S/N-00925 and as per drawing C61000-9. All applicable Airworthiness directives have been reviewed and actioned accordingly. Production Testing has been conducted IAW TR 424301 Issue 3. 1st Row Right Hand Double Seat Installed. Reference Weight & Balance for Latest Amendment. Current configuration empty weight is 7238.9 lbs and Center of Gravity is 213.55 inches. The following Production items remain open: - Interior Cabin Reading lights in-op, circuit breaker is collared and placarded in-op. - Interior not complete. Partially Installed. - F23 Chromate Primer and F24 Polyurethane topcoat not complete. - Flight crew has been briefed on all in-op systems Weight & Balance performed IAW AEROC V6.3.G.1 Section II. Electrical Load Analysis performed IAW AEROC V6.6.E.1 Section 2.2.1. Pitot/Static System Leak Checked in accordance with CAR Standard 571 appendix B Powerplants, propellers, hydraulics systems, fuel systems, electrical systems, and avionics systems are Fit & Safe for Flight. Aircraft is operating without a Certificate of Airworthiness. Operation shall be as per Production Flight permit RDIMS# 10741635. I have conducted an independent Inspection of the Engine Operation & Flight Control Systems for correct assembly, the presence of all safety and locking devices and correct sense of movement, through their full range of travel. Date: JUN 08 2015 Signature: [Signature] VA MFG 1151 The production activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and Industry recognized airworthiness requirements. Date: JUN 08 2015 Signature: [Signature] 70 VAQC Carried Out a Daily Pre-Flight Inspection as per Viking FORM-017-27 Control of the Aircraft is now Transferred to Flight Operations Production Release The Production Activities described above have been performed in accordance with the applicable airworthiness requirements and the aircraft is determined to be in a condition for safe operation. Date: JUN 08 2015 Signature: [Signature] VA MFG 1151 Aircraft is under the flight authority of specific purpose flight permit # RDIMS 10741635 The following operating limitations apply for this flight, Appendix: ☒ A) Production Test Flights ☐ B) Ferry Flights ☐ C) Customer Acceptance Flights ☐ Not Applicable		

2

Carried Out a Daily Pre-Flight Inspection as per Viking FORM-017-27 Control of the Aircraft is now Transferred to Flight Operations		Aircraft is under the flight authority of specific purpose flight permit # RDIMS <u>10741635</u>	
Production Release The Production Activities described above has been performed in accordance with the applicable airworthiness requirements and the aircraft is determined to be in a condition for safe operation.		The following operating limitations apply for this flight, Appendix:	
Date: <u>JUN 09 2015</u> Signature: <u>[Signature]</u>		☒ A) Production Test Flights ☐ B) Ferry Flights ☐ C) Customer Acceptance Flights ☐ Not Applicable	
A/C Reg: C-GNVA Model: DHC-6 400		S/N: 925	
TTSN 0.0 TCSN 0 Landings 0			
L/H Engine S/N: PCE-RB0886 TSN: 0 CSN: 0		R/H Engine S/N: PCE-RB0887 TSN: 0 CSN: 0	
L/H Prop S/N: BUA33359 TSN: 0		R/H Prop S/N: BUA33360 TSN: 0	
June 9, 2015			
Aircraft inspection completed as per SN-00925-2/850 Op 21.			
The production activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and Industry recognized airworthiness requirements.			
Date: <u>JUN 09 2015</u> Signature: <u>[Signature]</u>		VA MFG 1568	
Carried Out a Daily Pre-Flight Inspection as per Viking FORM-017-27 Control of the Aircraft is now Transferred to Flight Operations		Aircraft is under the flight authority of specific purpose flight permit # RDIMS <u>10741635</u>	
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Date: <u>JUN 09 2015</u> Signature: <u>[Signature]</u>		☒ A) Production Test Flights ☐ B) Ferry Flights ☐ C) Customer Acceptance Flights ☐ Not Applicable	

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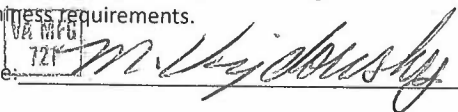
3

Date / /	Total air time	Total landings/cycles	Details of task	Signature	AME or AMO/ACA
2015/06/09	3.0	1/1	carry Forward		
A/C Reg: C-GNVA Model: DHC-6 400 S/N: 925 TTSN 3.0 TCSN 1 Landings 1 L/H Engine S/N: PCE-RB0886 TSN: 3.0 CSN: 1 R/H Engine S/N: PCE-RB0887 TSN: 3.0 CSN: 1 L/H Prop S/N: BUA33359 TSN: 3.0 R/H Prop S/N: BUA33360 TSN: 3.0 June 9, 2015 Aircraft inspection completed as per SN-00925-2/850 Op 22. The production activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and Industry recognized airworthiness requirements. Date: JUN 09 2015 Signature: [Signature] VA MFB 1151 Carried Out a Daily Pre-Flight Inspection as per Viking FORM-017-27 Control of the Aircraft is now Transferred to Flight Operations Production Release The Production Activities described above has been performed in accordance with the applicable airworthiness requirements and the aircraft is determined to be in a condition for safe operation. Date: JUN 09 2015 Signature: [Signature] VA MFB 1151 Aircraft is under the flight authority of specific purpose flight permit # RDIMS 10741635 The following operating limitations apply for this flight, Appendix: ☒ A) Production Test Flights ☐ B) Ferry Flights ☐ C) Customer Acceptance Flights ☐ Not Applicable					
2015/06/09	Hebert/Pasquali	CYK CYK	2356 0136 1.6 4.6 2 2	transcribed by Kern Heerema.	

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[illegible]

1

		Signature	AME or AMO/ACA
VIKING[®] VERSATILITY THAT WORKS			
Registration: C-GNVA	Model: DHC-6 400	S/N: 925	
Date: July 7th 2015	TTSN: 6.3 hrs	TCSN: 3	Landings: 3
-Flaps, Ailerons and all exterior Placard and Markings have been removed. -Aircraft exterior surfaces have been painted in accordance with Viking Engineering Drawing C6-2100 Rev NC, VPS 34.11 and NCM 50939 -Flaps and Ailerons have been re-installed in accordance with the applicable Viking Engineering Drawings as per WO: SN-00925-4/900 -All exterior Placards and Markings have been reapplied in accordance with Viking Production Flight Test Procedure FTP-6-400 Test card FTP-6GND Rev A Ref: WO ID: SN-00925-4/900 -Aircraft weight and balance amendment completed, new current configuration basic weight and centre of gravity is 7336.7 lbs at 214.38 inches ref: NCM 50939 - An EMMA 7 day interval check has been performed as per cards Routine A and Routine E dated: July 7th 2015 -Aircraft is operating without a Certificate of Airworthiness. I have conducted an independent inspection of the affected flight control systems for correct assembly, the presence of all safety locking devices and correct sense of movement, through the full range of travel.			
Signature: 		Date: JUL 07 2015	
The production activities described above have been performed in accordance with the applicable Viking Engineering data, Viking work order SN-00925/1, and industry recognized airworthiness requirements.			
Signature: 		Date: JUL 07 2015	

Control of the Aircraft is now Transferred to Flight Operations

Production Release

The Production Activities described above has been performed in accordance with the applicable airworthiness requirements and the aircraft is determined to be in a condition for safe operation.

Date: JUL 07 2015

Signature: 721 *W. J. [illegible]*

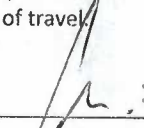
Aircraft is under the flight authority of specific purpose
flight permit # RDIMS 10741635.

The following operating limitations apply for this flight, Appendix:

- ☐ A) Production test flight ☒ B) Ferry Flights
- ☐ C) Customer Acceptance Flights ☐ D) Not Applicable

[illegible]

6

			
VERSATILITY THAT WORKS			
Registration: C-GNVA	Model: DHC-6 400	SN: 925	
Date: July 20th 2015	TTSN: 8.0 hrs.	TCSN: 4 Cycles	Landings: 4
-Interior installed as per the applicable Viking Engineering drawings -Interior lighting has been installed and tested as per TR424301 Section C -Compass system calibration carried out and checked serviceable -Compass correction cards installed as per WO ID: SN-00925-27/900 Dated: July 16 th 2015 -CVR reinstalled after intelligibility test. PN: 980-6023-002, SN: ARCVR-11623 -Function test carried out under WO ID: SN-00925-28/900 and found serviceable -Aileron trim servo removed and resealed under WO ID: SN-00925-25/900 -Weight and Balance was conducted IAW VPS 13.01 under WO ID: SN-00925- 30/900 -The new Basic Aircraft Weight and C of G is 7785.3 lbs at 1672402 AND 214.33 INCHES -An EMMA 7 day interval check has been performed as per cards Routine A and .81 Routine E dated : July 20 2015 -Aircraft is operating without a Certificate of Airworthiness			
I have conducted an independent inspection of the affected flight control system for correct assembly, the presence of all safety-locking devices and correct sense of movement, through the full range of travel.			
Signature: 		Date: JUL 20 2015	
The production activities described above have been performed in accordance with the applicable Viking Engineering data, Viking work order SN-00925/1, and industry recognized airworthiness requirements.			
Signature: 		Date: July 20 2015	

Section 1: Record of maintenance and elementary work

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Date / /	Total air time	Total landings/cycles	Details of task
22 JUL 2015	8.0	4 / 4	CARRIED FORWARD

LOGBOOK ENTRY

VIKING

TSN:	8.0	TCSN:	4	LDGS:	4
Pitot Static system check and calibration:			DATE:	03 JUN 2015	
Compass system swing and calibration:			DATE:	16 JUL 2015	
Last EMMA Routine A & E (7 day interval checks):			DATE:	20 JUL 2015	
Aircraft satisfactorily flight tested in accordance with Viking Air Limited Flight Test Procedure.			DATE:	14 JUL 2015	
Production Flight Test Release Certificate issued:			DATE:		
All Flight snags have been rectified as per applicable flight snag reports.					
Production activities have been performed in accordance with applicable Viking Air Limited Engineering Data. This aircraft is in a condition for safe operation.					
DATE:	22 JUL 2015	SIGNATURE:	S. VIGER		



STATEMENT OF CONFORMITY

VIKING

Aircraft

Manufacturer: Viking Air Limited
 Model: DHC-6 Twin Otter 400
 Type Certificate Data Sheet: TCCA A-82 Iss. 15
 Serial No.: 925

Engines

Left

Right

Manufacturer: Pratt & Whitney Canada
 Model: PT6A-34
 Type Certificate Data Sheet: TCCA E-6 Iss. 26
 Serial Numbers: PCE-RB0886

Propellers

Left

Right

Manufacturer: Hartzell Propeller Inc.
 Model: HC-B3TN-3DY/T10282NB
 Type Certificate Data Sheet: TCCA P-49 Iss. 2
 Serial Numbers: BUA33359

I hereby certify that I have test flown the aircraft described above, and that its performance, flying qualities, and functioning of controls, power plant, and landing gear are up to the standards of the Type aircraft.

Tested on: 14 JUL 2015

Signed: ANZS329
 (On Behalf of Viking Air Limited Flight Operations)

Date: 22 JUL 2015

I hereby certify that the aircraft described above, produced in accordance with the policies and procedures of the Viking Air Quality Manual (QM-001) under Transport Canada Approval Number 22-80, conforms to its approved type design and is in condition for safe operation.

Signed: SCOTT VIGER
 (On Behalf of Viking Air Limited Quality Control)

S. VIGER



Date: 22 JUL 2015

FORM-017-13 -- Statement of Conformity Technical Record Entry
WI-017-11

Rev. C

[illegible]

2

	Signature	AME or AMO/ACA
• Weight and Balance Amendment #2, approved by the Board of Directors on 12/15/2015		

Reference work order: MRO-SN0925-1

- Nav and chart database upload carried out as per P.S.M. 1-64-2 Chap. 45-00-00
- Routine EMMA 7 day A&E inspection carried out as per P.S.M. 1-6-7(1C)
- TB V6/0057 & V6/0069 completed on LH & RH engines. FCU rods replaced with new P/N: C6CE1398-7. MOD 6/2347 has been incorporated.
- Interior headliner panels removed and aircraft water tested. No defects found. Interior reinstalled as required.
- Installed new MLG blind clips and shock cords IAW REO6-32-V038.
- Installed S.O.O. 6325 (autopilot switch provisions) IAW Dwg. No. C6MK6325, Rev. NC.
- Installed S.O.O 6355 (autopilot servos) IAW Dwg. No. C6MK6355, Rev. NC.
 - Aileron servo P/N: 2R5310A, S/N: C3061
 - Aileron Capstan P/N: 291302A, S/N: A3978
 - Elevator Servo P/N: 2R5310A, S/N: C3216
 - Elevator Capstan P/N: 291302A, S/N: A3984
 - Rudder Servo P/N: 2R5310A, S/N: C3224
 - Rudder capstan P/N: 291302A, S/N: A3980
 - Pitch Trim Servo P/N: 065-0061-50. S/N: KS272A-48728

New Empty Weight 7821.31 Arm: 215.05 Moment: 1681975.07

Maintenance Release

Signature: [Signature] 22-80\429 AMO 22-80

Signature: M. J. [illegible] **VIKING AND** AMO 22-80
22-80-1111

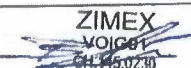
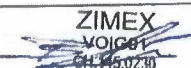
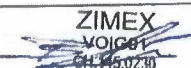
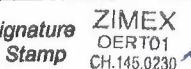
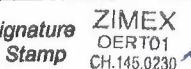
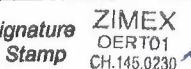
Certificate of Release to Service (CRS) i.a.w. Zimex MOE Appendix D

The work detailed below has been carried out on the above aircraft:

- EMMA INSPECTION #1 TO #12
 - CARD FUNCTION 125/250/500/1000 HRS OPERATIONAL TESTS
 - CARD CALENDAR 12 MTS INSPECTION
 - CARD CT-A-B1 - MAIN BATTERY CHECK
 - CARD CT-A-B2 - EMERGENCY LIGHT BATTERY CHECK
 - CARDS SP2: F2, J2, J3, L1, L2, L3, L4, L5, L6, N1, N2, N3, N4, N5, N6, N7, N8
 - CARD SP3 L1 TO L7 COLD WEATHER OPERATION KIT
 - ANNUAL RADIO CHECK AND NAV DATABASE UPDATE
 - ELT ANNUAL CHECK
 - REPLACED R/H RCR P/N: OFF/ON: A701D, S/N OFF: CT-70682, ON: CU-16694
 - REPLACED AILERON TRIM POTENTIOMETER, FUNCTIONAL CHECK OK.
 - MOD 6/2303 / SOO 6221 - COMPLETED PHASE 2A & AP INSTALLATION I.A.W. TBV6-00078
 - SB V6/0056 - INCORPORATION OF MOD 6/2380 - RED FUEL EMERGENCY SWITCHES
 - SB V6/0059 - GUARD ASSEMBLY, AIRSTAIR DOOR INNER HANDLE DESIGN IMPR.
 - LH/RH PROPELLER LUBRICATION
- WORK PERFORMED I.A.W. IRM PSM 1-6-7(IC), REV.18, AMM PSM 1-64-2 REV. 6,
EMM P/N: 3021242 REV.55 AND HARTZELL POM 139, REV. 14

Zimex / Customer WO Reference Numbers: #370020 / 370079

Zimex Aviation Ltd.		EASA #CH.145.0230 / TCCA #814-21	
Certificate of release to service for work performed above on Aircraft		Registration C-GNVA	
I certify that the work specified except as otherwise specified was carried out in accordance with EASA Regulations and in respect to that work the aircraft / aircraft component is considered ready for release service.			
Place: ALTENRHEIN	Work Order: 370020 / 370079		Signature & Stamp  ZIMEX VOIGT CH.145.0230
Date: 30.05.2017	Name: B.Voigt		

Date / /	Total air time	Total landings/cycles	Details of task																																																									
14-FEB-2020																																																												
31-MAY-2019																																																												
01-JUN-2017			 ZIMEX Zimex Aviation Ltd. Cherstrasse 4 CH-8152 Glattbrugg Switzerland Telefon ++41 44 815 53 53 Telefax ++41 44 815 53 21 EASA Part-145 approved CH.145.0230 Certificate of Release to Service (CRS) i.a.w. Zimex MOE Appendix D <table border="1"> <thead> <tr> <th></th> <th>Type</th> <th>S/N</th> <th>TSN</th> <th>TSO</th> <th>CSN</th> <th>CSO</th> </tr> </thead> <tbody> <tr> <td>Airframe</td> <td>DHC6-400</td> <td>925</td> <td>56.5</td> <td>--</td> <td>15</td> <td>--</td> </tr> <tr> <td>LH Engine</td> <td>PT6A-34</td> <td>PCE-RB0886</td> <td>56.5</td> <td>--</td> <td>15</td> <td>--</td> </tr> <tr> <td>RH Engine</td> <td>PT6A-34</td> <td>PCE-RB0887</td> <td>56.5</td> <td>--</td> <td>15</td> <td>--</td> </tr> <tr> <td>LH Propeller</td> <td>HC-B3TN-3DY</td> <td>BUA33359</td> <td>56.5</td> <td>--</td> <td>--</td> <td>--</td> </tr> <tr> <td>RH Propeller</td> <td>HC-B3TN-3DY</td> <td>BUA33360</td> <td>56.5</td> <td>--</td> <td>--</td> <td>--</td> </tr> </tbody> </table> The work detailed below has been carried out on the above aircraft: - REPLACED UPPER BEACON P/N: OFF/ON: 01-0771080-05, S/N OFF: 00520, ON: 00614 - REPLACED COCKPIT FIRE EXTINGUISHER P/N: OFF/ON: C352TS, S/N OFF: A-572406, ON: D-00397781 - WORK PERFORMED I.A.W. AMM PSM 1-63-2 Rev. 19. - Zimex / Customer WO Reference Numbers: #370020 <table border="1"> <tr> <td>Zimex Aviation Ltd.</td> <td>EASA #CH.145.0230 / TCCA #814-21</td> </tr> <tr> <td>Certificate of release to service for work performed above on Aircraft</td> <td>Registration C-GNVA</td> </tr> <tr> <td colspan="2">I certify that the work specified except as otherwise specified was carried out in accordance with EASA Regulations and in respect to that work the aircraft / aircraft component is considered ready for release to service.</td> </tr> <tr> <td>Place: ALTENRHEIN</td> <td>Work Order: 370020</td> </tr> <tr> <td>Date: 01.06.2017</td> <td>Name: B.Voigt</td> </tr> <tr> <td>Signature & Stamp</td> <td></td> </tr> </table>					Type	S/N	TSN	TSO	CSN	CSO	Airframe	DHC6-400	925	56.5	--	15	--	LH Engine	PT6A-34	PCE-RB0886	56.5	--	15	--	RH Engine	PT6A-34	PCE-RB0887	56.5	--	15	--	LH Propeller	HC-B3TN-3DY	BUA33359	56.5	--	--	--	RH Propeller	HC-B3TN-3DY	BUA33360	56.5	--	--	--	Zimex Aviation Ltd.	EASA #CH.145.0230 / TCCA #814-21	Certificate of release to service for work performed above on Aircraft	Registration C-GNVA	I certify that the work specified except as otherwise specified was carried out in accordance with EASA Regulations and in respect to that work the aircraft / aircraft component is considered ready for release to service.		Place: ALTENRHEIN	Work Order: 370020	Date: 01.06.2017	Name: B.Voigt	Signature & Stamp	
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Signature & Stamp																																																												

31-MAY-2019

 ZIMEX		Zimex Aviation Ltd. Cherstrasse 4 CH-8152 Glattbrugg Switzerland Telefon ++41 44 815 53 53 Telefax ++41 44 815 53 21 EASA Part-145 approved CH.145.0230
Certificate of Release to Service (CRS) i.a.w. Zimex MOE Appendix D		

	Type	S/N	TSN	TSO	CSN	CSO
Airframe	DHC6-400	925	57.6	--	16	--
LH Engine	PT6A-34	PCE-RB0886	57.6	--	16	--
RH Engine	PT6A-34	PCE-RB0887	57.6	--	16	--
LH Propeller	HC-B3TN-3DY	BUA33359	57.6	--	--	--
RH Propeller	HC-B3TN-3DY	BUA33360	57.6	--	--	--

The work detailed below has been carried out on the above aircraft:

- EMMA #13 TO #24 INSPECTION
- CARD FUNCTION 125 HOURS OPERATIONAL TESTS
- C/W AD'S: CF-79-22R2, CF-80-03R4, CF-2017-20, CF-2018-28, CF-2019-02
- C/W SB'S: 6/502, 6/521, V6/0022B, V6/0062
- EMMA CARD CAL 12 MTHS INSPECTION
- CARD SP2: B2, C7, D9, D10, E3, H1, J2, J3, L1 to L6, N1 to N9
- CARD SP3 L1-L7 COLD WEATHER OPERATIONS KIT INSPECTION
- CARD CT-A-B1 REPLACED MAIN BATTERY
- P/N OFF/ON: RG-380E/60L, S/N OFF: 40615204, S/N ON: 41002048
- FIRST AID KIT INSPECTION
- REPLACED HYDRAULIC OIL AND HYDRAULIC FILTER
- REPLACED NOSE WHEEL BEARINGS AND CUPS
- NAV & EGPWS DATABASE UPDATE
- PITOT STATIC SYSTEM LEAK CHECK
- ANNUAL AVIONIC EQUIPMENT CHECK
- EMPENNAGE REMOVAL / INSTALLATION (FOR PAINTING)
- WING FLIGHT CONTROLS REMOVAL / INSTALLATION (FOR PAINTING)
- C/W MOD 6/2417 INSTALLATION OF ELEVATOR POTENTIOMETER ACCESS PANEL
- LH / RH PROPELLER INSPECTION / LUBRICATION
- LH / RH ENGINE COMPRESSOR RECOVERY & CT WASH
- LH / RH ENGINE HOT SECTION BORESCOPE INSPECTION
- LH / RH ENGINE CHIP DETECTOR BRIDGING INSPECTION
- LH / RH ENGINE OIL CHANGE
- ENGINE GROUND RUN AND FUNCTIONAL / OPERATIONAL CHECKS CARRIED OUT
- WORK PERFORMED I.A.W. VIKING IRM PSM 1-6-7, REV. 20, AMM PSM 1-64-2, REV. 8, PWC EMM 3021242, REV.60 AND HARTZELL POM 139, REV.16

Zimex / Customer WO Reference Numbers: #370138

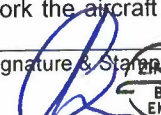
Zimex Aviation Ltd.	EASA #CH.145.0230 / TCCA #814-21
Certificate of release to service for work performed above on Aircraft	Registration C-GNVA
I certify that the work specified except as otherwise specified was carried out in accordance with EASA Regulations and in respect to that work the aircraft / aircraft component is considered ready for release to service.	
Place: ALTENRHEIN	Work Order: 370138
Date: 31.05.2019	Name: STEPHAN KLUG
Signature & Stamp	

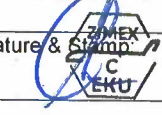
[illegible]

Date / /	Total air time	Total landings/cycles	Details of task			
			 Zimex Aviation Ltd. Cherstrasse 4 CH-8152 Glattbrugg Switzerland Telefon ++41 44 815 53 53 Telefax ++41 44 815 53 21 EASA Part-145 approved CH.145.0230			
Certificate of Release to Service (CRS) i.a.w. Zimex MOE Appendix D						
	Type	S/N	TSN	TSO	CSN	CSO
Airframe	DHC6-400	925	58.4	--	19	--
LH Engine	PT6A-34	PCE-RB0886	58.4	--	19	--
RH Engine	PT6A-34	PCE-RB0887	58.4	--	19	--
LH Propeller	HC-B3TN-3DY	BUA33359	58.4	--	--	--
RH Propeller	HC-B3TN-3DY	BUA33360	58.4	--	--	--
The work detailed below has been carried out on the above aircraft: - EMMA ROUTINE 7 DAY A & E INSPECTION - CARD SP2 N1 ESIS PERIODIC BATTERY CHECK - NAV DATABASE UPDATE WORK PERFORMED I.A.W. VIKING IRM PSM 1-6-7, REV. 20, AMM PSM 1-64-2, REV. 8, PWC EMM 3021242, REV. 62 AND HARTZELL POM 139, REV. 17 Zimex / Customer WO Reference Numbers: #400046						
Zimex Aviation Ltd.			EASA #CH.145.0230 / TCCA #814-21			
Certificate of release to service for work performed above on Aircraft			Registration C-GNVA			
I certify that the work specified except as otherwise specified was carried out in accordance with EASA Regulations and in respect to that work the aircraft / aircraft component is considered ready for release to service.						
Place: LSZR		Work Order: 400046		Signature & Stamp 		
Date: 11.05.2020		Name: ROBIN KARRER				

		Zimex Aviation Ltd. Cherstrasse 4 CH-8152 Glattbrugg Switzerland		Telefon ++41 44 815 53 53 Telefax ++41 44 815 53 21 EASA Part-145 approved CH.145.0230		
Certificate of Release to Service (CRS) i.a.w. Zimex MOE Appendix D						
	Type	S/N	TSN	TSO	CSN	CSO
Airframe	DHC6-400	925	58.9	--	21	--
LH Engine	PT6A-34	PCE-RB0886	58.9	--	21	--
RH Engine	PT6A-34	PCE-RB0887	58.9	--	21	--
LH Propeller	HC-B3TN-3DY	BUA33359	58.9	0.0	--	--
RH Propeller	HC-B3TN-3DY	BUA33360	58.9	0.0	--	--
The work detailed below has been carried out on the above aircraft: - EMMA #26 TO #36 INSPECTION - EMMA CARD CALENDAR 12 MTHS INSPECTION - CARD FUNCTION 125 / 250 / 500 / 1000 / 1500 / 2000 / 2400 / 3000 HOURS OPS TESTS - C/W AD: CF-72-06R5, CF-79-22R2, CF-80-03R4, CF-85-14, CF-2017-20, CF-2019-02 CF-2020-45R1, - C/W SB: 6/454, PART A, 6/474A, 6/502, 6/511A, 6/521C, V6/0022B, V6/0062, V6/0063A, V6/0067A, - CARD SP2: C1, C7, C8, C9, D9, D10, D12, E7, F4, G6, H1, J2, J3, J5, J6, L1-6, N1-9, O1 - CARD SP3 L1-L7 COLD WEATHER OPERATIONS KIT INSPECTION - AIRCRAFT WEIGHT AND BALANCE AND UPDATE OF EQUIPMENT LIST - CARD SP2 O1 MAIN BATTERY CAPACITY TEST P/N OFF/ON: RG-380E/60L, S/N OFF 40985213, S/N ON: 41173013 - EMERGENCY LIGHT BATTERY CHECK - FIRST AID KIT INSPECTION - ANNUAL ELT CHECK - REPLACED ELT BATTERY, P/N: 452-0133, S/N OFF: 265279-006, S/N ON: 392571-0133 - COCKPIT AND CABIN SEAT BELTS INSPECTION - LH / RH ENGINE FIRE EXTINGUISHER HYDROSTATIC TEST & SQUIB REPLACEMENT (PERFORMED BY HUGEN MFA) SQUIB P/N ON: AE13083-5, LH S/N ON: 4738, RH S/N ON: 4740 - [27-002 & 27-006] REPLACED ELEVATOR, RUDDER & AILERON FLIGHT CTRL CABLES - [27-007] REPLACED RUDDER & ELEVATOR PULLEYS AT STA 270.3 - REPLACED CVR ULB BATTERY, P/N ON: DK120/90, S/N ON: AT170663 - REPLACED FDR ULB BATTERY, P/N ON: DK120/90, S/N ON: AT170664 - REPLACED AGM CARD #1: P/N: 7036410-1902, S/N OFF: 1503C445, S/N ON: 1409B679 - REPLACED AGM CARD #2: P/N: 7036410-1902, S/N OFF: 1503C443, S/N ON: 08020677 - REPLACED ESIS UNIT, P/N: 501-1860-0403, S/N OFF: 5227, S/N ON: 5265 - REPLACED DAU #1, P/N ON: 85-302-1, S/N ON: AAR0245 - REPLACED DAU #2, P/N ON: 85-302-1, S/N ON: AAF2365 - ANNUAL AVIONIC EQUIPMENT CHECK - [53-006] FUSELAGE SIDE FRAME INSPECTION FOR CRACKS AT ST239.88 - CARD SP2 G6 - LH PROPELLER OVERHAUL P/N: HC-B3TN-3DY, S/N: BUA33359 - CARD SP2 G6 - RH PROPELLER OVERHAUL P/N: HC-B3TN-3DY, S/N: BUA33360 - LH / RH ENGINE FLUID HOSES INSPECTION - LH / RH ENGINE COMPRESSOR RECOVERY & CT WASH - LH / RH ENGINE HOT SECTION BORESCOPE INSPECTION - REPLACED LH ENGINE FUEL NOZZLES, P/N ON: MAN-34, S/N ON: ZIMB-FN34-003 - REPLACED RH ENGINE FULE NOZZLES, P/N ON: MAN-34, S/N ON: ZIMB-FN34-007 - LH / RH ENGINE BINDING OF FCU PUSH ROD AND BEARINGS INSPECTION - LH / RH ENGINE CHIP DETECTOR BRIDGING INSPECTION WORK PERFORMED I.A.W. VIKING IRM PSM 1-6-7, REV. 20, AMM PSM 1-64-2, REV. 8, PWC EMM 3021242, REV. 65 AND HARTZELL POM 139, REV. 19 Zimex / Customer WO Reference Numbers: #410079						
Zimex Aviation Ltd.			EASA #CH.145.0230 / TCCA #814-21			
Certificate of release to service for work performed above on Aircraft			Registration C-GNVA			
I certify that the work specified except as otherwise specified was carried out in accordance with EASA Regulations and in respect to that work the aircraft / aircraft component is considered ready for release to service.						
Place: LSZR		Work Order: 410079		Signature & Stamp 		
Date: 28.10.2021		Name: S. KLUG				

				Signature		AME or AMO/ACA	
		ZIMEX Zimex Aviation Ltd. Cherstrasse 4 CH-8152 Glattbrugg Switzerland		Telefon ++41 44 815 53 53 Telefax ++41 44 815 53 21 EASA Part-145 approved CH.145.0230			
Certificate of Release to Service (CRS) i.a.w. Zimex MOE Appendix D							
	Type	S/N	TSN	TSO	CSN	CSO	
Airframe	DHC6-400	925	59.3	--	22	--	
LH Engine	PT6A-34	PCE-RB0886	59.3	--	22	--	
RH Engine	PT6A-34	PCE-RB0887	59.3	--	22	--	
LH Propeller	HC-B3TN-3DY	BUA33359	59.3	0.4	--	--	
RH Propeller	HC-B3TN-3DY	BUA33360	59.3	0.4	--	--	
The work detailed below has been carried out on the above aircraft:							
- EMMA #37 TO #42 INSPECTION - EMMA CARD CALENDAR 12 MTHS INSPECTION - CARD FUNCTION 125 / 250 / 500 HOURS OPERATIONAL TESTS - CARD SP2 C1 - ELEVATOR QUADRANT AND SUPPORT BRACKET INSPECTION I.A.W. CF-72-06R5 - PART III AND SB 6/511A - CARD SP3 L1-L7 - COLD WEATHER OPERATIONS KIT INSPECTION - ELEVATOR AND RUDDER A/P BRIDLE CABLE RIGGING - CARD SP2 O1 - LEAD ACID MAIN BATTERY CAPACITY TEST - MOD 6/2314 - INDEPENDENT EXTERNAL & BATTERY POWER SWITCHES INST. I.A.W. TB V6/00073 - MOD 6/2315 - INSTALLATION OF DHC-6-400 SERIES AVIONIC START BATTERY I.A.W. TB V6/00125, P/N ON: RG-122-3, S/N ON: 70022307 - PS850B ESIS EMERGENCY POWER SUPPLY REPAIR, P/N: 501-1719-02, S/N: 80265 - FIRST AID KIT INSPECTION - CARD SP2 H1 - ANNUAL ELT CHECK (FAR 91.207) - COCKPIT AND CABIN SEAT BELTS INSPECTION - CARD SP2 J2 - HAND FIRE EXTINGUISHER WEIGHT CHECK (COCKPIT AND CABIN) - CARD SP2 J3 - LH / RH ENGINE FIRE EXTINGUISHER SYSTEM CHECK - PRIMARY FLIGHT CONTROL CABLE TENSION ADJUSTMENT AND RIGGING - CARD SP2 N1 - ESIS PERIODIC BATTERY CHECK - CARD SP2 N2 - ESIS BATTERY CAPACITY TEST - CARD SP2 N3 - CVR ULB PERIODIC CHECK - CARD SP2 N4 - FDR ULB PERIODIC CHECK - CARD SP2 N5 - CVR INTELLIGIBILITY TEST - CARD SP2 N6 - FDR CORRELATION CHECK - CARD SP2 N7 - ESIS CALIBRATION CHECK - CARD SP2 N8 - CVR FUNCTION TEST - REPLACED CUSTOM I/O MODULE P/N: 7036333-1902, S/N OFF: 15021533, S/N ON: 10010708 - ANNUAL AVIONIC EQUIPMENT CHECK - CARD SP2 L1 - STANDBY COMPASS SWING (FAR 23.1547) - CARD SP2 L2 - ADAHRS (KSG 7200) COMPASS ACCURACY VERIFICATION - NAV DATABASE UPDATE - [53-006] FUSELAGE SIDE FRAME INSPECTION AT ST239 FOR CRACKS (SB 6/521C) - LH / RH PROPELLER INSPECTION AND LUBRICATION - LH / RH ENGINE HOT SECTION BORESCOPE INSPECTION - LH / RH ENGINE COMPRESSOR RECOVERY & CT WASH - LH / RH ENGINE CHIP DETECTOR BRIDGING INSPECTION WORK PERFORMED I.A.W. VIKING IRM PSM 1-6-7, REV. 20, AMM PSM 1-64-2, REV. 8, PWC EMM 3021242, REV. 68 AND HARTZELL POM 139, REV. 19 Zimex / Customer WO Reference Numbers: #430055							
Zimex Aviation Ltd.				EASA #CH.145.0230 / TCCA #814-21			
Certificate of release to service for work performed above on Aircraft				Registration C-GNVA			
Certifies that except as otherwise specified, the described maintenance has been performed in accordance with the applicable airworthiness requirements and the approved MAG supplement and in respect to that work the aircraft is ready for release to service.							
Place: LSZR		Work Order: 430055		Signature & Stamp			
Date: 14.04.2023		Name: S. KLUG					

Date / /	Total air time	Total landings/cycles	Details of task
 Zimex Aviation Ltd. Cherstrasse 4 8152 Glattbrugg Switzerland Telephone +41 44 815 53 53 Telefax +41 44 815 53 21 EASA Part-145 Approved CH.145.0230			
Certificate of Release to Service (CRS) for Non-EASA Registered Aircraft			
Airframe	Type	S/N	TSN TSO CSN CSO
DHC-6-400	925	59.8	-- 23 --
LH Engine	PT6A-34	PCE-RB0886	59.8 -- 23 --
RH Engine	PT6A-34	PCE-RB0887	59.8 -- 23 --
LH Propeller	HC-B3TN-3DY	BUA33359	59.8 0.9 -- --
RH Propeller	HC-B3TN-3DY	BUA33360	59.8 0.9 -- --
The work detailed below has been carried out on the aircraft above:			
- EMMA #43 TO #48 INSPECTION (STANDARD D-CHECK) - EMMA ROUTINE 7 DAY A & E INSPECTION - REGISTRATION CHANGE (C-GNVA TO UP-DH601) - MOD 6/2315 - INSTALLATION OF OVERHEAD PANEL - REPROGRAMMED ELT AND TRANSPONDER (C-GNVA TO UP-DH601) - CARD SP2 N1 - ESIS PERIODIC BATTERY CHECK			
Work performed in accordance with: VIKING IRM PSM 1-6-7(IC), REV. 20, AMM PSM 1-64-2, REV. 8, EMM NO. 3021242, REV. 70, HARTZELL POM NO. 139, REV. 20			
For further details refer to Zimex Aviation Ltd. / customer work order # 430285			
Zimex Aviation Ltd.			
Certificate of release to service for work performed on aircraft			UP-DH601
Certifies that the work specified, except as otherwise specified, was carried out in accordance with the Article 48 of the Law 339-IV of the Republic of Kazakhstan and in recognition of the organisation's EASA Part-145 approval, and in respect to that work the aircraft is considered ready for release to service.			
Place: LSZR	Work Order: 430285	Signature & Stamp	
Date: 30 Aug. 2023	Name: E. KUNZ	  ZMX-MX-128	

							Signature	AME or AMO/ACA
 Zimex Aviation Ltd. Cherstrasse 4 8152 Glattbrugg Switzerland Telephone +41 44 815 53 53 Telefax +41 44 815 53 21 EASA Part-145 Approved CH.145.0230								
Certificate of Release to Service (CRS) for Non-EASA Registered Aircraft								
Airframe	Type	S/N	TSN	TSO	CSN	CSO		
DHC-6-400	925	59.8	--	--	23	--		
LH Engine	PT6A-34	PCE-RB0886	59.8	--	23	--		
RH Engine	PT6A-34	PCE-RB0887	59.8	--	23	--		
LH Propeller	HC-B3TN-3DY	BUA33359	59.8	0.9	--	--		
RH Propeller	HC-B3TN-3DY	BUA33360	59.8	0.9	--	--		
The work detailed below has been carried out on the aircraft above:								
- EMMA #01 TO #06 INSPECTION, - EMMA ROUTINE 7 DAY A & E INSPECTION - EMMA CARD CAL 12 MTHS INSPECTION - CARD SP2: C7, D9, D10, H1, J2, J3, L1-L6, N1-N9, O1 - CARD SP3: L1-L7 COLD WEATHER OPERATION KIT INSPECTION - C/W AD: CF-79-22R2, CF-80-03R4, CF-2019-02, CF-2021-42 - C/W SB: 6/502, 6/521C, V6/0063A - HYDRAULIC PUMP RELAY VISUAL INSPECTION AND CB REPLACEMENT - FIRST AID KIT INSPECTION - COCKPIT AND CABIN SEAT BELTS INSPECTION - ANNUAL ELT CHECK (FAR 91.207) (EASA SIB 2019-09 R1) - ANNUAL AVIONIC EQUIPMENT CHECK - REPLACED KSG7200 ADAHRS WITH UPGRADED UNIT P/N OFF: 065-00188-5103, S/N OFF: KSG7200-B2512 P/N ON: 065-00188-5104, S/N ON: KSG7200-B1317 - PITOT STATIC SYSTEM LEAK CHECK (FAR 91.411) - [53-006] FUSELAGE SIDE FRAME INSPECTION AT ST239 FOR CRACKS (SB 6/521C) - LH / RH PROPELLER INSPECTION / LUBRICATION - LH / RH ENGINE COMPRESSOR RECOVERY & CT WASH - CF-2021-42 (PT III) - LH / RH ENGINE FCU PUSH ROD END BEARINGS INSPECTION - LH / RH ENGINE CHIP DETECTOR BRIDGING INSPECTION								
Work performed in accordance with: VIKING IRM PSM 1-6-7(IC), REV. 20, AMM PSM 1-64-2, REV. 9, EMM NO. 3021242, REV. 73.1, HARTZELL POM NO. 139, REV. 22								
For further details refer to Zimex Aviation Ltd. / customer work order # 440295								
Zimex Aviation Ltd.								
Certificate of release to service for work performed on aircraft							UP-DH601	
Certifies that the work specified, except as otherwise specified, was carried out in accordance with the Article 48 of the Law 339-IV of the Republic of Kazakhstan and in recognition of the organisation's EASA Part-145 approval, and in respect to that work the aircraft is considered ready for release to service.								
Place: LSZR		Work Order: 440295		Signature & Stamp				
Date: 25 Oct. 2024		Name: E. KUNZ		  ZMX-MX-128				

Reference (e.g., AD #)	Date applicable / /	Compliance threshold	Compliance interval	Brief description or requirement	Entered by	Date entered / /																																																																																																																																																										
AIRWORTHINESS DIRECTIVE COMPLIANCE REPORT COMPONENT: DHC-6 Series 400 Airframe SERIAL NUMBER: 925 <table><thead><tr><th>A.D. Number</th><th>A.D. Description</th><th>Method of Compliance</th><th>N/A*</th><th>C*</th><th>R*</th><th>Interval</th></tr></thead><tbody><tr><td>CF-68-05</td><td>Additional balance weights introduced to assure ailerons meet aerodynamic requirements.</td><td>MOD 6/1104 incorporated during manufacture.</td><td></td><td>X</td><td></td><td></td></tr><tr><td>CF-68-09</td><td>Reinforcement to left hand rear Spar to preclude possibility of elevator trim tab flutter.</td><td>MODs 6/1120, 6/1135 and 6/1142 incorporated during manufacture.</td><td></td><td>X</td><td></td><td></td></tr><tr><td>CF-69-20</td><td>To detect loose or missing Jo-bolts in wing struts which have accumulated 150 hours or more time in service.</td><td>Not applicable due to wing strut part number.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-70-10</td><td>To provide adequate protection for particular circuit wires.</td><td>Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-71-07</td><td>Improved sump plate flapper valve seal - To preclude separation of seal from flapper valve in cells 4 and 5.</td><td>MOD 6/1406 incorporated during manufacture.</td><td></td><td>X</td><td></td><td></td></tr><tr><td>CF-71-11</td><td>To prevent possible explosions in bays 9 and 10 due to the operation of electrical components, deactivate the J.B. Systems Air Conditioner.</td><td>Not applicable air-conditioning system not installed.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-71-12</td><td>Slow blow fuses Type 313 replaced with fast blow fuses Type 312 - To improve protection of AC power on aircraft.</td><td>Not applicable due to aircraft serial number. MOD 6/2028 incorporated during manufacture.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-72-06R4</td><td>Reinforced lower torque tube - To eliminate 50-hour inspection required by F.A.A. A.D. Redesignated Elevator/Rudder pulley bracket - to prevent buckling and cracking due to high ground gust loading.</td><td>MODs 6/1262, and 6/1433 incorporated during manufacture.</td><td></td><td></td><td>X</td><td>Next inspection of elevator quadrant due at 400 flight hours.</td></tr><tr><td>CF-73-03R1</td><td>Wing Rear Spar - Change of material provide improved stress corrosion cracking properties.</td><td>MOD 6/1301 incorporated during manufacture.</td><td></td><td>X</td><td></td><td></td></tr><tr><td>CF-73-04</td><td>Propeller Autofeather - To introduce autofeather control on PT6A-20 engines</td><td>Not applicable due to aircraft serial number. MOD 6/2028 incorporated during manufacture.</td><td>X</td><td></td><td></td><td></td></tr></tbody></table> VERSATILITY THAT WORKS FORM-017-01 - Airworthiness Directive Compliance Report WI-017-13 Rev.C Page 1 of 6 AIRWORTHINESS DIRECTIVE COMPLIANCE REPORT COMPONENT: DHC-6 Series 400 Airframe SERIAL NUMBER: 925 <table><thead><tr><th>A.D. Number</th><th>A.D. Description</th><th>Method of Compliance</th><th>N/A*</th><th>C*</th><th>R*</th><th>Interval</th></tr></thead><tbody><tr><td>CF-73-06R2</td><td>Overhead console control quadrant - Introduction of improved material - To preclude cracking.</td><td>MOD 6/1895 incorporated during manufacture.</td><td></td><td>X</td><td></td><td></td></tr><tr><td>CF-73-09</td><td>Propeller Autofeather - Pressure switches wiring modified to overcome erratic operation of single pole pressure switches.</td><td>Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-73-11</td><td>Float attachment bolts - Special inspection and replacement</td><td>Not applicable floats not installed.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-73-12</td><td>Side Seat Rail attachment - Special inspection of rivets and replacement as required.</td><td>Not applicable inward facing seats not installed.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-74-04</td><td>Horizontal Stabilizer Rear Spar - Special inspection and repair.</td><td>Not applicable by aircraft serial number.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-74-07</td><td>Electrical Power Cable below cockpit - Repositioning of terminal block 17.</td><td>Not applicable by aircraft serial number.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-75-06</td><td>Slow blow fuses Type 313 replaced with fast blow fuses Type 312 - To improve protection of AC power on aircraft.</td><td>Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-75-11</td><td>To preclude the possibility of partial or total electrical failure due to contact welding of reverse current relays, Hartman Part A700AAP or A700AP.</td><td>Not applicable new part number A701D installed. MOD 6/2028 incorporated during manufacture.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-75-14</td><td>Floats - Applies to all de Havilland DHC-6 Model aircraft fitted with Canadian Aircraft Products Models 12000, 12000A or 12000B floats modified for water bombing operation</td><td>Not applicable floats not installed.</td><td>X</td><td></td><td></td><td></td></tr><tr><td>CF-76-21</td><td>Lighting protection - Wing tip, fincap, nose fuselage (SFAR 23 requirement).</td><td>Not applicable by aircraft serial number. MOD 6/1513 installed during manufacture.</td><td>X</td><td></td><td></td><td></td></tr></tbody></table> VERSATILITY THAT WORKS FORM-017-01 - Airworthiness Directive Compliance Report WI-017-13 Rev.C Page 2 of 6							A.D. Number	A.D. Description	Method of Compliance	N/A*	C*	R*	Interval	CF-68-05	Additional balance weights introduced to assure ailerons meet aerodynamic requirements.	MOD 6/1104 incorporated during manufacture.		X			CF-68-09	Reinforcement to left hand rear Spar to preclude possibility of elevator trim tab flutter.	MODs 6/1120, 6/1135 and 6/1142 incorporated during manufacture.		X			CF-69-20	To detect loose or missing Jo-bolts in wing struts which have accumulated 150 hours or more time in service.	Not applicable due to wing strut part number.	X				CF-70-10	To provide adequate protection for particular circuit wires.	Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.	X				CF-71-07	Improved sump plate flapper valve seal - To preclude separation of seal from flapper valve in cells 4 and 5.	MOD 6/1406 incorporated during manufacture.		X			CF-71-11	To prevent possible explosions in bays 9 and 10 due to the operation of electrical components, deactivate the J.B. Systems Air Conditioner.	Not applicable air-conditioning system not installed.	X				CF-71-12	Slow blow fuses Type 313 replaced with fast blow fuses Type 312 - To improve protection of AC power on aircraft.	Not applicable due to aircraft serial number. MOD 6/2028 incorporated during manufacture.	X				CF-72-06R4	Reinforced lower torque tube - To eliminate 50-hour inspection required by F.A.A. A.D. Redesignated Elevator/Rudder pulley bracket - to prevent buckling and cracking due to high ground gust loading.	MODs 6/1262, and 6/1433 incorporated during manufacture.			X	Next inspection of elevator quadrant due at 400 flight hours.	CF-73-03R1	Wing Rear Spar - Change of material provide improved stress corrosion cracking properties.	MOD 6/1301 incorporated during manufacture.		X			CF-73-04	Propeller Autofeather - To introduce autofeather control on PT6A-20 engines	Not applicable due to aircraft serial number. MOD 6/2028 incorporated during manufacture.	X				A.D. Number	A.D. Description	Method of Compliance	N/A*	C*	R*	Interval	CF-73-06R2	Overhead console control quadrant - Introduction of improved material - To preclude cracking.	MOD 6/1895 incorporated during manufacture.		X			CF-73-09	Propeller Autofeather - Pressure switches wiring modified to overcome erratic operation of single pole pressure switches.	Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.	X				CF-73-11	Float attachment bolts - Special inspection and replacement	Not applicable floats not installed.	X				CF-73-12	Side Seat Rail attachment - Special inspection of rivets and replacement as required.	Not applicable inward facing seats not installed.	X				CF-74-04	Horizontal Stabilizer Rear Spar - Special inspection and repair.	Not applicable by aircraft serial number.	X				CF-74-07	Electrical Power Cable below cockpit - Repositioning of terminal block 17.	Not applicable by aircraft serial number.	X				CF-75-06	Slow blow fuses Type 313 replaced with fast blow fuses Type 312 - To improve protection of AC power on aircraft.	Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.	X				CF-75-11	To preclude the possibility of partial or total electrical failure due to contact welding of reverse current relays, Hartman Part A700AAP or A700AP.	Not applicable new part number A701D installed. MOD 6/2028 incorporated during manufacture.	X				CF-75-14	Floats - Applies to all de Havilland DHC-6 Model aircraft fitted with Canadian Aircraft Products Models 12000, 12000A or 12000B floats modified for water bombing operation	Not applicable floats not installed.	X				CF-76-21	Lighting protection - Wing tip, fincap, nose fuselage (SFAR 23 requirement).	Not applicable by aircraft serial number. MOD 6/1513 installed during manufacture.	X			
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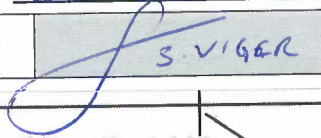
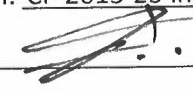
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JUN 05 2019

Section 3: Record of applicable airworthiness directives, airworthiness limitation, and equivalent mandatory requirements

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Reference (e.g., AD #)	Date applicable / /	Compliance threshold	Compliance interval	Brief description or requirement	Entered by	Date entered / /																					
As of this date, I certify that this aircraft complies with all applicable Airworthiness Directives. TCCA through: <u>CF-2015-22</u> DATE: <u>22 JUL 2015</u> SIGNATURE: <u>S. VIGER</u>  																											
Mandatory Service Bulletin Compliance Report  <table border="1"> <thead> <tr> <th>COMPONENT</th> <th>SERIAL NUMBER</th> <th>S.B. Number</th> <th>S.B. Description</th> <th>Signature</th> <th>Stamp</th> <th>Date</th> </tr> </thead> <tbody> <tr> <td>Airframe (DHC-6 Twin Otter 400)</td> <td>925</td> <td></td> <td></td> <td></td> <td></td> <td></td> </tr> <tr> <td colspan="2">All mandatory service bulletins that apply to the DHC-6 Series 400 at the date of manufacture have been incorporated.</td> <td></td> <td></td> <td><u>S. VIGER</u> SCOTT VIGER</td> <td></td> <td>22 JUL 2015</td> </tr> </tbody> </table> VERSATILITY THAT WORKS FORM-017-07 - Mandatory Service Bulletin Compliance Report Rev. B Page 1 of 1							COMPONENT	SERIAL NUMBER	S.B. Number	S.B. Description	Signature	Stamp	Date	Airframe (DHC-6 Twin Otter 400)	925						All mandatory service bulletins that apply to the DHC-6 Series 400 at the date of manufacture have been incorporated.				<u>S. VIGER</u> SCOTT VIGER		22 JUL 2015
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As of this date, I certify that this aircraft complies with all applicable TCCA Airworthiness Directives. TCCA Airworthiness Directives through: <u>CF-2015-23 including CF-2014-20R1 and CF-2015-16R1</u> Authorized Signatory, AMO 22-80 <u></u> Stamp  Date <u>AUG 07 2015</u>																											
  Date: 13 May 2016 Model: DHC-6-400 A/C: C-GNVA S/N: 925 TAT: 8.0 hrs, 4 starts, with 4 landings: Reference work order: MRO-SN0925-1 The following work was carried out at Viking Air Limited. - Installed S.O.O. 6325 (autopilot switch provisions) IAW Dwg. No. C6MK6325, Rev. NC. - Installed S.O.O. 6355 (autopilot servos) IAW Dwg. No. C6MK6355, Rev. NC. - Aileron servo P/N: 2R5310A, S/N: C3061 - Aileron Capstan P/N: 291302A, S/N: A3978 - Elevator Servo P/N: 2R5310A, S/N: C3216 - Elevator Capstan P/N: 291302A, S/N: A3984 - Rudder Servo P/N: 2R5310A, S/N: C3224 - Rudder capstan P/N: 291302A, S/N: A3980 - Pitch Trim Servo P/N: 065-0061-50, S/N: KS272A-48728 EIE. <u>MR</u> 																											
					<u>M. Reardon</u>	 13 MAY 2016																					
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Date embodied / /	STC / RDA / PDA	Brief description of change	ICA changes (x / ✓)	Sched. change (x / ✓)	Description of changes to ICA and reference to applicable documents	Entered by																																																																																																																																																																																																																																																																																																																																							
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Modifications denoted with an asterisk (*) are partially embodied: <table><tr><td>6/1003</td><td>6/1004</td><td>6/1005</td><td>6/1007</td><td>6/1008</td><td>6/1010</td><td>6/1011</td><td>6/1012</td><td>6/1013</td></tr><tr><td>6/1015</td><td>6/1021</td><td>6/1022</td><td>6/1024</td><td>6/1025</td><td>6/1028</td><td>6/1029</td><td>6/1030</td><td>6/1031</td></tr><tr><td>6/1032</td><td>6/1036</td><td>6/1037</td><td>6/1038</td><td>6/1039</td><td>6/1042</td><td>6/1043</td><td>6/1044</td><td>6/1048</td></tr><tr><td>6/1050</td><td>6/1056</td><td>6/1059</td><td>6/1060</td><td>6/1062</td><td>6/1063</td><td>6/1065</td><td>6/1066</td><td>6/1069</td></tr><tr><td>6/1070</td><td>6/1071</td><td>6/1076</td><td>6/1077</td><td>6/1079</td><td>6/1084</td><td>6/1085</td><td>6/1086</td><td>6/1088</td></tr><tr><td>6/1089</td><td>6/1090</td><td>6/1094</td><td>6/1095</td><td>6/1098</td><td>6/1100</td><td>6/1101</td><td>6/1102</td><td>6/1103</td></tr><tr><td>6/1104</td><td>6/1107</td><td>6/1108</td><td>6/1110</td><td>6/1112</td><td>6/1113</td><td>6/1114</td><td>6/1115</td><td>6/1117</td></tr><tr><td>6/1120</td><td>6/1121</td><td>6/1122</td><td>6/1123</td><td>6/1124</td><td>6/1127</td><td>6/1128</td><td>6/1129</td><td>6/1135</td></tr><tr><td>6/1142</td><td>6/1143</td><td>6/1153</td><td>6/1157</td><td>6/1159</td><td>6/1160</td><td>6/1161</td><td>6/1162</td><td>6/1164</td></tr><tr><td>6/1171</td><td>6/1172</td><td>6/1173</td><td>6/1174</td><td>6/1176</td><td>6/1179</td><td>6/1181</td><td>6/1182</td><td>6/1184</td></tr><tr><td>6/1187</td><td>6/1189</td><td>6/1193</td><td>6/1194</td><td>6/1195</td><td>6/1198</td><td>6/1201</td><td>6/1203</td><td>6/1204</td></tr><tr><td>6/1206</td><td>6/1207</td><td>6/1209</td><td>6/1211</td><td>6/1213</td><td>6/1214</td><td>6/1215</td><td>6/1219</td><td>6/1222</td></tr><tr><td>6/1223</td><td>6/1224</td><td>6/1225</td><td>6/1226</td><td>6/1227</td><td>6/1228</td><td>6/1229</td><td>6/1230</td><td>6/1231</td></tr><tr><td>6/1232</td><td>6/1233</td><td>6/1235</td><td>6/1245</td><td>6/1247</td><td>6/1248</td><td>6/1250</td><td>6/1253</td><td>6/1255</td></tr><tr><td>6/1256</td><td>6/1257</td><td>6/1258</td><td>6/1260</td><td>6/1262</td><td>6/1263</td><td>6/1264</td><td>6/1268</td><td>6/1269</td></tr><tr><td>6/1270</td><td>6/1272</td><td>6/1274</td><td>6/1277</td><td>6/1279</td><td>6/1280</td><td>6/1280</td><td>6/1293</td><td>6/1294</td></tr><tr><td>6/1296</td><td>6/1298</td><td>6/1299</td><td>6/1301</td><td>6/1302</td><td>6/1306</td><td>6/1308</td><td>6/1309</td><td>6/1314</td></tr><tr><td>6/1315</td><td>6/1316</td><td>6/1319</td><td>6/1322</td><td>6/1325</td><td>6/1326</td><td>6/1328</td><td>6/1329</td><td>6/1333</td></tr><tr><td>6/1334</td><td>6/1335</td><td>6/1336</td><td>6/1337</td><td>6/1338</td><td>6/1339</td><td>6/1340</td><td>6/1341</td><td>6/1342</td></tr><tr><td>6/1344</td><td>6/1346</td><td>6/1348</td><td>6/1349</td><td>6/1351</td><td>6/1352</td><td>6/1354</td><td>6/1356</td><td>6/1359</td></tr><tr><td>6/1360</td><td>6/1361</td><td>6/1363</td><td>6/1365</td><td>6/1367</td><td>6/1368</td><td>6/1375</td><td>6/1376</td><td>6/1393</td></tr><tr><td>6/1395</td><td>6/1396</td><td>6/1397</td><td>6/1403</td><td>6/1406</td><td>6/1407</td><td>6/1412</td><td>6/1415</td><td>6/1417</td></tr><tr><td>6/1419</td><td>6/1428</td><td>6/1427</td><td>6/1430</td><td>6/1433</td><td>6/1435</td><td>6/1439</td><td>6/1440</td><td>6/1448</td></tr><tr><td>6/1450</td><td>6/1451</td><td>6/1453</td><td>6/1457</td><td>6/1458</td><td>6/1461</td><td>6/1462</td><td>6/1463</td><td>6/1468</td></tr><tr><td>6/1470</td><td>6/1473</td><td>6/1476</td><td>6/1477</td><td>6/1486</td><td>6/1487</td><td>6/1488</td><td>6/1489</td><td>6/1491</td></tr><tr><td>6/1494</td><td>6/1496</td><td>6/1497</td><td>6/1499</td><td>6/1501</td><td>6/1502</td><td>6/1505</td><td>6/1508</td><td>6/1513</td></tr><tr><td>6/1515</td><td>6/1523</td><td>6/1529</td><td>6/1530</td><td>6/1537</td><td>6/1538</td><td>6/1540</td><td>6/1541</td><td>6/1542</td></tr><tr><td>6/1543</td><td>6/1547</td><td>6/1548</td><td>6/1554</td><td>6/1556</td><td>6/1557</td><td>6/1561</td><td>6/1562</td><td>6/1563</td></tr><tr><td>6/1566</td><td>6/1567</td><td>6/1570</td><td>6/1580</td><td>6/1581</td><td>6/1582</td><td>6/1583</td><td>6/1585</td><td>6/1590</td></tr><tr><td>6/1591</td><td>6/1593</td><td>6/1594</td><td>6/1596</td><td>6/1597</td><td>6/1599</td><td>6/1601</td><td>6/1603</td><td>6/1604</td></tr><tr><td>6/1606</td><td>6/1609</td><td>6/1612</td><td>6/1614</td><td>6/1615</td><td>6/1616</td><td>6/1618</td><td>6/1620</td><td>6/1633</td></tr></table> VERSATILITY THAT WORKS FORM-017-10 - Modification Summary WI-017-11 Rev - B Page 1 of 3 Approved  MODIFICATION SUMMARY  The following approved modifications have been incorporated to aircraft serial number 925 during production. 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6/1003	6/1004	6/1005	6/1007	6/1008	6/1010	6/1011	6/1012	6/1013																																																																																																																																																																																																																																																																																																																																					
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